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(ISO 9001 CERTIFIED) AIR TRAFFIC MANAGEMENT DIVISION CIVIL AVIATION DEPARTMENT HONG KONG INTERNATIONAL AIRPORT

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TERMINATION OF ALLOCATION OF SLOTS TO NON-CHAPTER 4 EQUIVALENT AIRCRAFT OPERATING AT NIGHT TO/FROM HONG KONG INTERNATIONAL AIRPORT

1. Introduction

- 1.1 Since July 2002, all subsonic jet aircraft landing or taking off in Hong Kong have been required to meet the noise standards stipulated in Volume I, Part II, Chapter 3 of Annex 16 to the Convention on International Civil Aviation ("Chapter 3").
- For the purpose of improving the local noise environment, operations of Marginally Compliant Chapter 3 (MCC3) ¹ aircraft have been restricted into the Hong Kong International Airport (HKIA) since the Summer Season of 2014. The Civil Aviation Department (CAD) has not been allocating slots to flights which are proposed to be operated by MCC3 aircraft to land or take off in Hong Kong throughout the whole day.
- To further reduce the noise impact to the local communities while giving more preparation time 1.3 to aircraft operators, the CAD will tighten the restriction and will not allocate slots to Non-Chapter 4 Equivalent aircraft in a progressive manner. Commencing from the slot allocation for the Summer Season of 2019², no slot will be allocated to Non-Chapter 4 Equivalent aircraft operating between 1400 UTC and 2259 UTC ("Prohibited Period").

2. Non-Chapter 4 Equivalent Aircraft

- 2.1 Chapter 4 aircraft refer to civil subsonic jet aircraft that meet the noise standards stipulated in Volume I, Part II, Chapter 4 of Annex 16 to the Convention on International Civil Aviation. The Chapter 4 noise standards that are more stringent than the Chapter 3 noise standards applied to aircraft which are type-certificated between 2006 and 2017.
- For the purpose of implementing the abovementioned restriction, Chapter 3 aircraft that have 2.2 met the criteria stipulated in Section 4.4 of the aforementioned ICAO Annex 16 provisions are considered as Chapter 4 Equivalent aircraft, despite the fact that they might not have been re-certificated to Chapter 4 noise standards by the relevant authority or manufacturer. Those which are UNABLE to meet such criteria are then considered as Non-Chapter 4 Equivalent aircraft. The methodology in ascertaining whether an aircraft meets the definition of Non-Chapter 4 Equivalent aircraft is shown in the Attachment hereto. Some commonly known Non-Chapter 4 Equivalent aircraft are the older models of A321 and B737-300 and B737-400.

¹ Marginally Compliant Chapter 3 (MCC3) aircraft are defined as subsonic jet aircraft which comply with the noise standards stipulated in Volume I, Part II, Chapter 3 of the Annex 16 to the Convention on International Civil Aviation by a cumulative margin of not more than 5 EPNdB. Some commonly known aircraft types, include Boeing B727, Boeing B737 (-100 and -200 series), Boeing B747 (-100, -200, -300 and SP series), McDonnell Douglas DC-10 etc., are considered as MCC3 aircraft. The methodology in ascertaining whether an aircraft meets this definition is shown in the Attachment hereto.

² Start of Season on 31 March 2019.

2.3 In theory, the MCC3 aircraft are also Non-Chapter 4 Equivalent aircraft. For the avoidance of doubt, the MCC3 aircraft restriction will continue to apply throughout the whole day as per paragraph 1.2.

3. Pre-assessment of Commonly Known MCC3 Aircraft or Non-Chapter 4 Equivalent Aircraft

- 3.1 As the certificated noise performance of individual aircraft may vary depending on its engineering configuration, it is possible that an aircraft, though its type is one of the commonly known MCC3 or Non-Chapter 4 Equivalent aircraft types, may indeed be fall outside the respective definitions. Therefore, to facilitate the smooth implementation of the restrictions, an operator is recommended to carry out a self-assessment of the noise performance of its aircraft as early as possible in accordance with the methodology in the Attachment before submitting slot requests.
- 3.2 If an operator has reason(s) to believe that its aircraft may NOT be classified as MCC3 or Non-Chapter 4 Equivalent aircraft, it is advised to request the Environmental Management Office of the CAD to conduct an official noise assessment of the aircraft. The operator is required to submit the data necessary for such assessment which include registration mark, certificated maximum take-off mass, number of engines and certificated noise levels of each aircraft. Operators should provide documentary evidence on the data submitted. The assessment by the Environmental Management Office will be registration mark-specific. Each individual aircraft, which is classified as Non-MCC3 or Chapter 4 Equivalent aircraft after the assessment, will be issued a Pre-assessment Non-MCC3 Reference Number or a Pre-assessment Chapter 4 Equivalent Reference Number, as the case may be.
- 3.3 The contact details of the Environmental Management Office are:

Environmental Management Office Civil Aviation Department CAD Headquarters 1 Tung Fai Road Hong Kong International Airport Lantau, Hong Kong

Telephone: (852) 2769 6969 Fax: (852) 2326 3654

Email: aircraftnoise@cad.gov.hk

- 4. Requesting Slots for Commonly Known MCC3 Aircraft (for operation throughout the whole day) or Non-Chapter 4 Equivalent Aircraft (for operation during the Prohibited Period)
- 4.1 For a slot request proposing the use of an aircraft type commonly known as MCC3 aircraft for operations throughout the whole day, the CAD Hong Kong Schedule Coordination Office (HKSCO) will apply the MCC3 restriction as per paragraph 1.2, unless the slot applicant has included in the Slot Clearance Request/Reply (SCR) message under Supplementary Information (SI) the corresponding Pre-assessment Non-MCC3 Reference Number(s) of the aircraft to be used in the proposed operation. In the absence of a valid Pre-assessment Non-MCC3 Reference Number, no slot will be confirmed or offered by HKSCO.
- 4.2 For a slot request proposing the use of an aircraft type commonly known as Non-Chapter 4 Equivalent aircraft for operations during the Prohibited Period, subject to paragraph 4.3 below, the HKSCO of the CAD will apply the restriction as per paragraph 1.3, unless the slot applicant

has included in the SCR message under SI the corresponding Pre-assessment Chapter 4 Equivalent Reference Number(s) of the aircraft to be used in the proposed operation. In the absence of a valid Pre-assessment Chapter 4 Equivalent Reference Number, no slot will be confirmed or offered by HKSCO.

- 4.3 The requirement on the inclusion of a valid Pre-assessment Chapter 4 Equivalent Reference Number in the SCR message for slot requests proposing the use of aircraft types A321, B737-300 or B737-400 during the Prohibited Period will generally be waived. However, the slot applicant shall ensure that the operating aircraft is Chapter 4 Equivalent. The CAD will take appropriate action against any repeated and intentional misuse of this arrangement.
- 4.4 Should the slot applicant be applying for a noise assessment with the Environmental Management Office at the same time as submitting the slot request, the SCR message should be emailed to HKSCO at <a href="https://hksco.org/hksco.or

5. Dispensation for the Use of MCC3 Aircraft or Non-Chapter 4 Equivalent Aircraft within the Prohibited Period in Unforeseen Circumstances

- 5.1 If an operator allocated with a slot needs to operate an MCC3 aircraft due to unforeseen circumstances (such as aircraft deployment needs), the CAD will be prepared to consider such a request on a case-by-case basis but the operator concerned will be required to provide a written report to the Environmental Management Office with the justifications within 96 hours after the flight is flown.
- 5.2 If an operator with an allocated slot within the Prohibited Period needs to operate a Non-Chapter 4 Equivalent aircraft due to unforeseen circumstances (such as aircraft deployment needs, unavoidable operational delay, etc.), the CAD will be prepared to consider such a request on a case-by-case basis but the operator concerned will be required to provide a written report to the Environmental Management Office with the justifications within 96 hours after the flight is flown.
- 5.3 The CAD will take appropriate action against any failure to provide the written report with the justifications, repeated and intentional misuse of the above arrangements.

6. Dispensation for the Use of MCC3 Aircraft in Insurmountable Circumstances

6.1 If an operator needs to operate an MCC3 aircraft due to insurmountable operational circumstances, the operator is required to submit its request with strong justifications supported with documentary evidence and information in relation to the aircraft to the Environmental Management Office for consideration. It must be stressed that the CAD will only grant exemption for operation of an MCC3 aircraft under very exceptional circumstances.

7.1	Non-Chapter 4 Equivalent and MCC3 aircraft operated as State aircraft, or for emergency medical and humanitarian purposes are exempted from the above restriction.
8.	AIC 18/18 is hereby superseded.
	This Circular is issued for information, guidance and necessary action
	by direction of the Director-General of Civil Aviation Victor LIU

7.

Exempted Aircraft

Example 1 – B737-200

Aircraft : B737-200 Maximum Take-off Weight (MTOW) : 52,400 kg

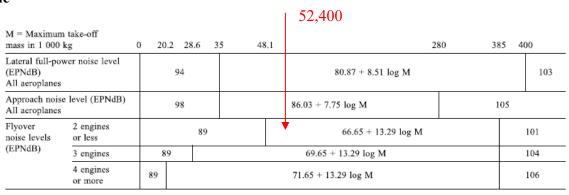
Number of engines : 2

Aircraft Certificated Noise Levels : - Lateral (L1) 96.2 (EPNdB)

- Approach (A1) 98.4 (EPNdB) - Flyover (F1) 87.7 (EPNdB)

Applicable standards in ICAO Annex 16, Vol. I, Chapter 3, 3.4

Table



Calculation of ICAO Limits

a) Lateral Full Power (L)

 $L = 80.87 + 8.51 \log (52,400/1,000)$

 $= \underline{95.5 \text{ (EPNdB)}}$

b) Approach (A)

 $A = 86.03 + 7.75 \log (52,400/1,000)$

= 99.4 (EPNdB)

c) Flyover (F)

As the aircraft has 2 engines and with MTOW 52,400 kg, the equation 66.65+13.29 log M applies.

 $F = 66.65+13.29 \log (52,400/1,000)$ = 89.5 (EPNdB)

Assessment of compliance

Criteria 1

Non-MCC3 aircraft are defined as civil aircraft that meet Chapter 3 standards by a cumulative margin of more than 5 EPNdB.

Calculation:

Sum [L+A+F] : [95.5+99.4+89.5] = 284.4 (EPNdB) Sum [L1+A1+F1] : [96.2+98.4+87.7] = 282.3 (EPNdB)

Difference = 2.1 (EPNdB) i.e. not > 5

Criteria 1 has <u>not</u> been met.

Conclusion

Since Criteria 1 cannot be met, the aircraft is a **Marginally Compliant Chapter 3 aircraft**.

Example 2 – B737-300

Aircraft : B737-300 Maximum Take-off Weight (MTOW) : 58,967 kg

Number of engines : 2

Aircraft Certificated Noise Levels : - Lateral (L1) 90.3 (EPNdB)

- Approach (A1) 100.0 (EPNdB) - Flyover (F1) 85.5 (EPNdB)

Applicable standards in ICAO Annex 16, Vol. I, Chapter 4, 4.4

Table

						58,967			
M = Maximum take-off mass in 1 000 kg 0 20.2 28.6 35 48.1							280	385	400
Lateral full-pov (EPNdB) All aeroplanes		94			80.87 + 8.51 log M			103	
Approach noise All aeroplanes		98			86.03 + 7.75 log M		105	•	
Flyover noise levels	2 engines or less		89			66.65 + 13.29 log M	•		101
(EPNdB)	3 engines	8	89		69.65 + 13.29 log M			104	
	4 engines or more	89				71.65 + 13.29 log M			106

Calculation of ICAO Limits

a) Lateral Full Power (L)

 $L = 80.87+8.51 \log (58,967/1,000)$ = 95.9 (EPNdB)

b) Approach (A)

A = 86.03+7.75 log (58,967/1,000) = 99.8 (EPNdB)

c) Flyover (F)

As the aircraft has 2 engines and with MTOW 58,967 kg, the equation 66.65+13.29 log M applies.

 $F = 66.65+13.29 \log (58,967/1,000)$ = 90.2 (EPNdB)

Assessment of compliance

Criteria 1

Para 4.4.1 The maximum permitted noise levels are defined in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, and shall not be exceeded at any of the measurement points.

Illustration:

L1 < L (Lateral Full Power)

 $A1 \le A$ (Approach)

F1 < F (Flyover)

Calculation:

Criteria 1 has <u>not</u> been met.

Criteria 2

Para 4.4.1.1 The sum of the differences at all three measurement points between the maximum noise levels and the maximum permitted noise levels specified in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, shall not be less than 10 EPNdB.

Illustration:

Sum [L+A+F] – Sum $[L1+A1+F1] \ge 10$

Calculation:

Sum [L+A+F] : [95.9+99.8+90.2] = 285.9 (EPNdB) Sum [L1+A1+F1] : [90.3+100.0+85.5] = 275.8 (EPNdB)

Difference = 10.1 (EPNdB) i.e. ≥ 10

Criteria 2 has been met.

Criteria 3

Para 4.4.1.2 The sum of the differences at any two measurement points between the maximum noise levels and the corresponding maximum permitted noise levels specified in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, shall not be less than 2 EPNdB.

Illustration:

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\begin{array}{ll} Sum \ [L+A] - Sum \ [L1+A1] & \geq 2 \\ Sum \ [A+F] - Sum \ [A1+F1] & \geq 2 \\ Sum \ [L+F] - Sum \ [L1+F1] & \geq 2 \end{array}
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Calculation:

Criteria 3 has been met.

Conclusion

Since criteria 1 cannot be met, the aircraft is a Non-Chapter 4 Equivalent aircraft.

Example 3 – A321-211

Aircraft : A321-211 Maximum Take-off Weight (MTOW) : 89,000 kg

Number of engines : 2

Aircraft Certificated Noise Levels : - Lateral (L1) 97.7 (EPNdB)

- Approach (A1) 96.9 (EPNdB)

- Flyover (F1) 88.0 (EPNdB)

Applicable standards in ICAO Annex 16, Vol. I, Chapter 4, 4.4

Table

					١	89,000				
M = Maximum mass in 1 000		0 20	.2 28.6	35 48		280	385	400		
Lateral full-pov (EPNdB) All aeroplanes		94			80.87 + 8.51 log M			103		
Approach noise All aeroplanes		98			86.03 + 7.75 log M		105	•		
Flyover noise levels	2 engines or less		89			66.65 + 13.29 log M	•		101	
(EPNdB)	3 engines	8	9			69.65 + 13.29 log M			104	
	4 engines or more	89	·			71.65 + 13.29 log M			106	

Calculation of ICAO Limits

a) Lateral Full Power (L)

 $L = 80.87 + 8.51 \log (89,000/1,000)$

= 97.5 (EPNdB)

b) Approach (A)

 $A = 86.03 + 7.75 \log (89,000/1,000)$

= 101.1 (EPNdB)

c) Flyover (F)

As the aircraft has 2 engines and with MTOW 89,000 kg, the equation 66.65+13.29 log M applies.

 $F = 66.65+13.29 \log (89,000/1,000)$ = 92.6 (EPNdB)

Assessment of compliance

Criteria 1

Para 4.4.1 The maximum permitted noise levels are defined in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, and shall not be exceeded at any of the measurement points.

Illustration:

L1 < L (Lateral Full Power)

A1 < A (Approach)

F1 < F (Flyover)

Calculation:

L1	< L	:	L1 = 97.7 (EPNdB)	L = 97.5 (EPNdB)	i.e. L1 is not \leq L
A 1	< A	:	A1 = 96.9 (EPNdB)	A = 101.1 (EPNdB)	i.e. $A1 < A$
F1	< F	:	F1 = 88.0 (EPNdB)	F = 92.6 (EPNdB)	i.e. $F1 < F$

Criteria 1 has not been met.

Criteria 2

Para 4.4.1.1 The sum of the differences at all three measurement points between the maximum noise levels and the maximum permitted noise levels specified in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, shall not be less than 10 EPNdB.

Illustration:

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Sum [L+A+F] - Sum [L1+A1+F1] \ge 10
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Calculation:

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Sum [L+A+F] : [97.5+101.1+92.6] = 291.2 (EPNdB)
Sum [L1+A1+F1] : [97.7+96.9+88.0] = 282.6 (EPNdB)
Difference = 8.6 (EPNdB) i.e. less than 10
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Criteria 2 has not been met.

Criteria 3

Para 4.4.1.2 The sum of the differences at any two measurement points between the maximum noise levels and the corresponding maximum permitted noise levels specified in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, shall not be less than 2 EPNdB.

Illustration:

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\begin{array}{ll} Sum \; [L+A] - Sum \; [L1+A1] & \geq 2 \\ Sum \; [A+F] - Sum \; [A1+F1] & \geq 2 \\ Sum \; [L+F] - Sum \; [L1+F1] & \geq 2 \end{array}
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Calculation:

Criteria 3 has been met.

Conclusion

Since both Criteria 1 and 2 cannot be met, the aircraft is a **Non-Chapter 4 Equivalent aircraft**.

Example 4 – A330-343

Aircraft : A330-343 Maximum Take-off Weight (MTOW) : 233,000 kg

Number of engines : 2

Aircraft Certificated Noise Levels : - Lateral (L1) 97.4 (EPNdB)

-Approach (A1) 97.0 (EPNdB) - Flyover (F1) 90.7 (EPNdB)

Applicable standards in ICAO Annex 16, Vol. I, Chapter 4, 4.4

Table

C						233,000			
M = Maximun mass in 1 000		0 20	.2 28.6	35 48	8.1	2	80 385	400	
Lateral full-po (EPNdB) All aeroplanes		94	80.87 + 8.51		log M		103		
Approach nois All aeroplanes		98		86.03 + 7.75 log M		105	105		
Flyover noise levels	2 engines or less		89)	66.65 + 13		101		
(EPNdB)	3 engines	8	9		69.65 + 13.29 log M				
	4 engines or more	89			71.65 + 13.29 log M			106	

Calculation of ICAO Limits

a) Lateral Full Power (L)

 $L = 80.87 + 8.51 \log (233,000/1,000)$

= <u>101.0 (EPNdB)</u>

b) Approach (A)

 $A = 86.03 + 7.75 \log (233,000/1,000)$

= 104.4 (EPNdB)

c) Flyover (F)

As the aircraft has 2 engines and with MTOW 233,000 kg, the equation 66.65+13.29 log M applies.

 $F = 66.65+13.29 \log (233,000/1,000)$ = 98.1 (EPNdB)

Assessment of compliance

Criteria 1

Para 4.4.1 The maximum permitted noise levels are defined in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, and shall not be exceeded at any of the measurement points.

Illustration:

L1 < L (Lateral Full Power)

 $A1 \le A$ (Approach)

 $F1 \le F$ (Flyover)

Calculation:

L1	< L	:	L1 = 97.4 (EPNdB)	L = 101.0 (EPNdB)	i.e. $L1 < L$
A 1	< A	:	A1 = 97.0 (EPNdB)	A = 104.4 (EPNdB)	i.e. $A1 \le A$
F1	< F	:	F1 = 90.7 (EPNdB)	F = 98.1 (EPNdB)	i.e. $F1 < F$

Criteria 1 has been met.

Criteria 2

Para 4.4.1.1 The sum of the differences at all three measurement points between the maximum noise levels and the maximum permitted noise levels specified in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, shall not be less than 10 EPNdB.

Illustration:

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Sum [L+A+F] - Sum [L1+A1+F1] \ge 10
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Calculation:

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\begin{array}{lll} Sum \ [L+A+F] & : & [101.0+104.4+98.1] & = 303.5 \ (EPNdB) \\ Sum \ [L1+A1+F1] & : & [97.4+97.0+90.7] & = 285.1 \ (EPNdB) \\ & & Difference & = 18.4 \ (EPNdB) \ i.e. \ge 10 \\ \end{array}
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Criteria 2 has been met.

Criteria 3

Para 4.4.1.2 The sum of the differences at any two measurement points between the maximum noise levels and the corresponding maximum permitted noise levels specified in Chapter 3, 3.4.1.1, 3.4.1.2 and 3.4.1.3, shall not be less than 2 EPNdB.

Illustration:

```
\begin{array}{ll} Sum \ [L+A] - Sum \ [L1+A1] & \geq 2 \\ Sum \ [A+F] - Sum \ [A1+F1] & \geq 2 \\ Sum \ [L+F] - Sum \ [L1+F1] & \geq 2 \end{array}
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Calculation:

Criteria 3 has been met.

Conclusion

Since all three criteria have been met, the aircraft is a **Chapter 4 Equivalent aircraft**.